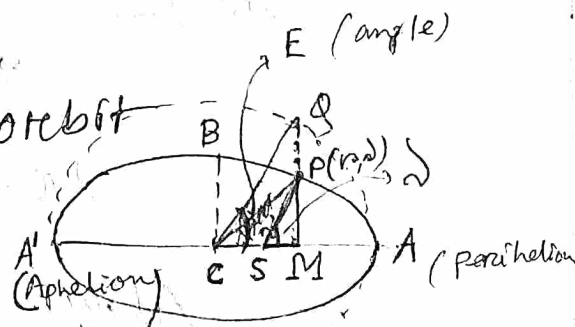


Keplars Equation

Let us consider an elliptic orbit whose semi major axis is a and ~~and~~ on auxiliary circle of radius ~~of radius~~ a , that has been circumscribed about ellipse.

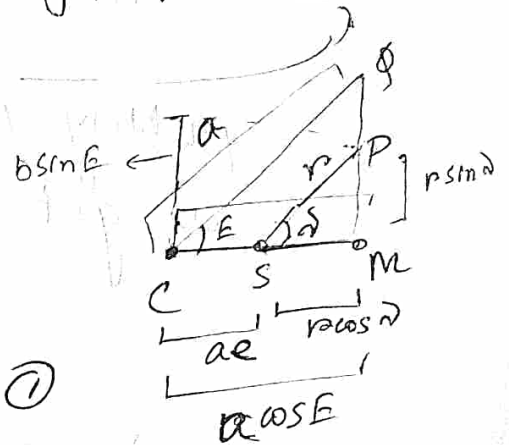


Let ϕ be the position of the planet after

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time "t" -

- after starting from perihelion position. Let S be the position of the Sun, also let λ be the true anomaly, E the Eccentric anomaly and m the mean anomaly. then



$$r \cos \lambda = a \cos E - ae \quad \rightarrow (1)$$

$$r \sin \lambda = b \sin E$$

$$= a \sqrt{1-e^2} \sin E \quad \rightarrow (2)$$

$$\therefore b^2 = a^2(1-e^2)$$

Then ~~the~~ squaring (1) and (2), then adding, we get

$$r^2 = a^2(1 - e \cos E)^2$$

$$\Rightarrow r = a(1 - e \cos E) \quad \rightarrow (3)$$

Also, $m = n(t - \tau) = nt$

$$= \frac{2\pi t}{T} \quad \rightarrow (4) \quad \tau = 0$$

where T is the periodic time

$$T = \frac{2\pi}{\sqrt{\mu}} a^{3/2} \rightarrow (5)$$

Now from (3), $\dot{r} = ae \sin E \cdot \dot{E} \rightarrow (6)$

The eqn of the orbit is given by $\frac{r}{r_0} = 1 + e \cos \nu$

$$\therefore + \frac{1}{r^2} \dot{r} = + e \sin \nu \dot{\nu}$$

$$\Rightarrow \dot{r} = \frac{e \sin \nu}{r} (r^2 \dot{\nu})$$

$$= \frac{\sin \nu}{l} h \rightarrow (7)$$

(6) and (7) gives $[\because r^2 \dot{\nu} = h]$

$$ae \sin E \cdot \dot{E} = \frac{e \sin \nu}{l} h$$

$$\Rightarrow a \sin E \cdot \dot{E} = \frac{\sin \nu}{l} \sqrt{\mu l}$$

$$= \frac{\sin \nu}{b^2/a} \sqrt{\mu \frac{b^2}{a}} \quad \Bigg| \quad l = \frac{b^2}{a}$$

$$= \sin \nu \sqrt{\frac{a\mu}{b^2}}$$

$$= \sin \nu \sqrt{\frac{a\mu}{a^2(1-e^2)}}$$

$$\Rightarrow \sin \nu \sqrt{\frac{\mu}{a(1-e^2)}}$$

$$\Rightarrow \dot{E} = \frac{\sqrt{\mu}}{\sqrt{a(1-e^2)}} \sin \nu \frac{1}{a \sin E}$$

$$= \frac{\sqrt{\mu}}{a} \sin \nu \frac{1}{\sqrt{1-e^2} \sin E}$$

$$= \sqrt{\frac{\mu}{a}} \frac{\sin \nu}{\mu \sin E} \quad [\text{using } \textcircled{2}]$$

$$= \sqrt{\frac{\mu}{a}} \cdot \frac{1}{\mu}$$

$$= \sqrt{\frac{\mu}{a}} \cdot \frac{1}{a(1-e \cos E)} \quad [\text{using } \textcircled{3}]$$

$$\Rightarrow (1-e \cos E) dE = \sqrt{\frac{\mu}{a^3}} dt$$

Integrating, $E - e \sin E = \sqrt{\frac{\mu}{a^3}} t + C$

But at the perihelion, $E=0$, $t=0$

$$C=0$$

$$E - e \sin E = \sqrt{\frac{\mu}{a^3}} t \quad \rightarrow \textcircled{4}$$

Now $\textcircled{5} \Rightarrow \sqrt{\mu} = \frac{2\pi}{T} a^{3/2}$

$$= \phi \pi a^{3/2}$$

$$\Rightarrow \sqrt{\frac{\mu}{a^3}} = n \rightarrow (9)$$

Using (9), (8) becomes

$$E - e \sin E = nt = m$$

$$\boxed{E - e \sin E = m}$$

This eqn is known as the Kepler's Equation.